

Flight Hardware

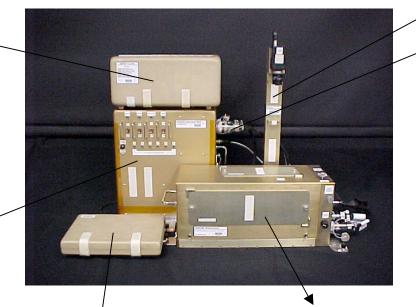
SHERE



Tool Box



Interface Box



Keyboard/Mouse

Camera Arm

▼ Cables (QTY=7)

Main Hardware:

Mass: 29.1 kg Volume: 0.100 m³

Fluid Module Stowage Tray

(not pictured):

Mass: 7.3 kg Volume: 0.012 m³



Rheometer



Flight Hardware

SHERE



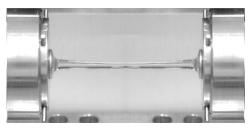
Fluid Module with Outer Shell Removed



Fluid Module Interior



Fluid Deployment



Polymeric Liquid Bridge In Microgravity



Fluid Module Stowage Tray (Contains 25 Fluid Modules)



Thermal Carrier (20° C for 24 hours)

Commercial Generic Bioprocessing Apparatus (CGBA)



Operations Scenario SHERE

- Hardware Installation
- Hardware Turn-on
- Hardware Checkout
- Fluid Sample Installation
- Experiment Execution
 - Test point selected
 - Preshear and stretched (automatic exponential velocity profile)
 - Stretch stopped at 194mm length
 - Fluid allowed to relax and break in half
 - Re-position slider to starting position
 - Fluid column is recombined (if possible)
- Fluid Sample Removal (prepare next sample or testing complete)
- Hardware Shutdown
- Hardware Removal



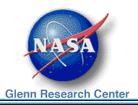
Constraints and Issues SHERE

- Replacing BXF (10A) and working towards H/W turnover in April, 2007
- No slack in schedule with the following taking place:
 - Integrated testing with MSG (scheduled for March 12-16, 2007)
 - Off-gas testing at MSFC (scheduled for March 19-23, 2007)
 - EMI TIA waiver in progress (expected completion February, 2007)
- Work currently in progress:
 - Software completion
 - Verification testing
 - Final calibrations
 - Miscellaneous H/W items
- Late stowage of Fluid Modules (see next slide)
- Crew training:
 - No Increment 16 operations
 - Increment 17 operations possible



Fluid Module Late Stowage Rationale SHERE

- On-orbit storage time limited by Fluid Module deployment ease.
 - Test fluid begins to act like glue over long periods of time.
 - Aged fluid causes binding of the sliding seals of the Fluid Module.
 - Data shows very difficult deployment after 1.4 and 2.5 years of storage.
 - Data shows easy deployment after 2.5 and 4.4 months of storage.
 - No data for gap between 4.4 months and 1.4 years of storage.
 - 6 months chosen as conservative limit to on-orbit testing.
 - Experiment to resolve issue is impractical given schedule and H/W availability.
- On-orbit storage time is also based on PI recommended material stability criteria
- Node-2 stowage is a risk to both FM deployment and leak integrity.
- L-2 weeks late stowage has been a SHERE requirement from inception.
- FM Stowage Tray can be launched separately from main H/W at a later time.



Progress Launch Options SHERE



- Not possible to launch Fluid Modules on Progress
 - Late stowage item
 - Thermal extremes (freezing and heating) a concern
- Possible to launch main H/W using Progress
- Issues would involve re-qualification to higher extremes
 - Higher vibration and shock loads
 - Larger thermal temperature extremes
- Unknown if would need to re-test H/W or if can be waived